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Volume 85 Technical rept ISAF BIOENVIRONMENTAL NOISE DATA HANDBOOK Volume 85, OV-10A Aircraft, Near and Far-Field Noise Justus F. Rose _____ Robert G. / Powell Approved for public release; distribution unlimited. AEROSPACE MEDICAL RESEARCH LABORATORY

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FOR THE COMMANDER

HENNING E. VON GIERKE

Biodynamics and Bionics Division Aerospace Medical Research Laboratory

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interference level, perceived noise level, and limiting times for total daily exposure of personnel with and without standard Air Force ear protectors. Far-field data measured at 19 locations are normalized to standard meteorological conditions and extrapolated from 75-8000 meters to derive sets of equal-value contours for these same seven acoustic measures as functions of angle and distance from the source. Refer to Volume 1 of this handbook, JUSAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application, AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc.

PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 723104, Measurement and Prediction of Noise Environments of Air Force Operations.

The authors gratefully acknowledge Mr. John Cole for his assistance in preparing this report, Lt. Col Donald Gasaway of the USAFSAM/NGEA, Brooks AFB, TX for providing near-field data, Mr. Keith Kettler, Mr. Henry Mohlman and Mr. David Eilerman of the University of Dayton for assistance in the mechanics of data processing, and Ms. Norma Peachey and Mr. Mike Patterson for assistance in typing and preparation of the graphics.

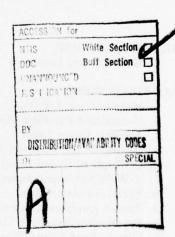


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INTRODUCTION

The USAF OV-10A is a forward air control or observation/strike reconnaissance aircraft powered by one each T76-G-10, 12 turboprop engines. The aircraft was manufactured by North American Rockwell and the engines by the Garrett Corporation.

This volume provides measured and extrapolated data defining bioacoustic environments produced by this aircraft during ground runup operations. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with ground runups of the OV-10A aircraft. The measured data presented in this volume were acquired by the Aerospace Medical Research Laboratory (AMRL), Wright-Patterson AFB, OH, and the USAF School of Aerospace Medicine (USAFSAM), Brooks AFB, TX.

This volume is one of a series published by the AMRL under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations an in surrounding communities by operations of Air Force aircraft and ground support equipment. The far-field, community-type, noise data in the handbook describe the noise produced during ground operations of aircraft, ground support equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discussed the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Volume 2 provides a method and data for adjusting the handbook's far-field noise data, which are for standard meteorological conditions (15 C temperature, 70% relative humidity, 0.760 meters Hg barometric pressure), to derive comparable data for other meteorological conditions. Refer to Volumes 1 and 2 (reference 1 and 2) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., inflight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published an is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of each updated index.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; AUTOVON 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application, AMRL-TR-75-50 (1) Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 2: Procedure to Evaluate Effects of Non-standard Meteorological Conditions on Far-Field Noise, AMRL-TR-75-50 (2), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

NEAR-FIELD NOISE

MEASUREMENTS

USAFSAM acquired near-field noise data on the OV-10A aircraft during ground runup operations of its turboprop engines (Reference 3). For these tests the aircraft was at Eglin Air Force Base, Hurlburt Field, FL. Table 1 lists the four engine-power conditions and near-field locations. The ground-crew chief selected power conditions and near-field locations generally used during routine maintenance or engine runup for preflight checks.

At each near-field location a test engineer randomly moved a hand-held microphone in and around each location, probing all areas where a crew member's head would normally be located. During this test he recorded a 15-20 second noise sample on magnetic tape at each location. During analysis of each sample, he determined the octave band root-mean-square sound pressure levels. Figure 1 shows the seven near-field locations where ground crew are usually located for maintenance and/or preflight checkout operations. Estimates of noise levels at other locations are difficult in the near-field since the noise source is spatially distributed, i.e., not a point source. The noise levels at near-field locations can vary widely depending upon relative distances from each noise source (intake noise, exhaust noise, panel resonances, internal engine noise through the engine wall, etc.).

Table 1 lists the numeric/alphabetic designators used on the data pages in this report to identify the measurement locations and test conditions. For example, the designator 1/A ground crew location 1 and test condition A.

RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced by the OV-10A aircraft at the seven ground crew locations. This table includes the overall and octave band levels. From these data one can calculate the variety of measures given in Table 3, which are widely used to assess the effects of noise or personnel and their performance.

Gasaway, Donald C., Noise Associated With Operation of Air Force OV-10A Aircraft, SAM-TR-70-51, USAF School of Aerospace Medicine, Brooks AFB, Texas, 1970.

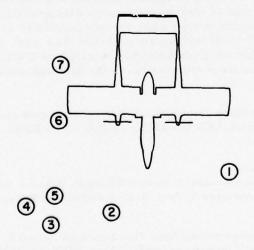


Figure 1. Near-Field Measurement Locations on a Taxiway at Hurlburt Field, FL

TABLE 1

MEASUREMENT LOCATIONS AND TEST CONDITIONS

FOR NEAR-FIELD NOISE MEASUREMENTS

OV-10A Aircraft, Ground Runups Hurlburt Field, FL

Ground Crew Location*

1		Left Side, 60 degrees at 22 ft.
2		Right Side, 20 degrees at 18 ft.
3		Right Side, 45 degrees at 32 ft.
4		Right Side, 60 degrees at 35 ft.
5		Right Side, 60 degrees at 23 ft.
6		Right Side, 90 degrees at 22 ft.
7		Right Side, 135 degrees at 26 ft.
Aircraft	Engine Operation	
A		Left Engine Idle
В		Both Engines Idle
c		Both Engines Taxi
D		Both Engines Taxi, High RPM

 $^{{\}bf *Locations}$ are relative to the intersection of the aircraft's centerline and the propellers' plane.

FAR-FIELD NOISE

MEASUREMENTS

AMRL acquired all far-field data during a 1-hour period, thus keeping similar meteorological conditions throughout the test. Figure 2 shows the ground runup pad, ground cover, aircraft orientation and 19 microphone measurement sites on each of two semicircles. The center of the 76 meter radius semicircle used in surveying the T76-G-10, 12 engines was on the ground directly below the intersection of the aircraft's centerline and the plane passing through both engines' propeller planes.

Table 4 provides cockpit readouts of engine characteristics (% RPM and torque) for each power setting used in the far-field tests. Also listed in this table are the surface meteorological conditions during data acquisition.

All 19 microphone measurement sites are in the acoustic far-field of their respective source where the sound wave-fronts spherically diverge and the noise source may be regarded as a point source.

A portable microphone/tape recorder system was used to sequentially record 5 to 10 seconds of noise at each far-field location. The microphone was hand-held 1.7 meters (5-1/2 feet) above the ground and pointed at the source (0° angle of incidence). These samples were then time-integrated to derive a root-mean-square sound pressure level.

RESULTS

Table 5 lists the overall and 1/3 octave band SPLmeasured at the far-field locations under meteorological conditions at the time of the test. Data in all other figures and tables are based on these levels. These data were normalized to 200 meters distance and standard meteorological conditions (15 C temperature, 70% relative humidity, 0.760 Hg barometric pressure) and used to derive the graphic data in Figure 3 which provides a compact summary of the far-field noise characteristics of the OV-10A aircraft in a standard format.

Figure 4 and Table 6 present two basic acoustic measures, the acoustic power levels and the directivity index, respectively. The acoustic power level describes the power radiated by the source as a function of frequency. The directivity index is a standard acoustical engineering measure that describes the geometric way in which the source radiates this power as a function of both frequency and angle from source. These basic source measures are primarily of interest for acoustical engineers and noise generation/control specialists.

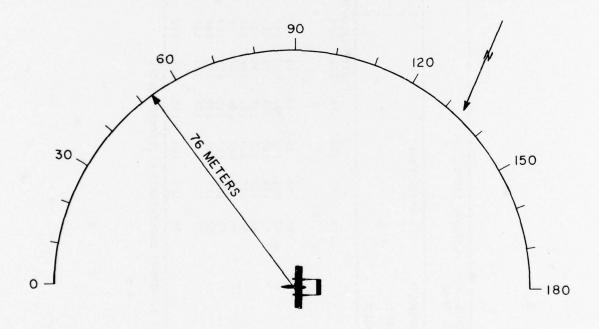
Estimates of the noise characteristics for intermediate power settings (e.g., 80% RPM) and/or different number of engines operating (e.g., single engine) can be determined as explained in Volume 1 of this handbook.

Figures 5 through 11 are sets of equal noise contours describing seven different measures of noise as a function of angle and distance from the source for standard day meteorology. They are respectively, overall sound pressure level, C-weighted sound level, A-weighted sound level, perceived noise level, speech interference level, permissible exposure times for personnel and octave band sound pressure levels.

Data excessively influenced by spurious background/electronic noise were eliminated from all figures and tables. No data are presented at the 180-degree location for the idle and military power settings because of turbulent air flow behind the aircraft. Typically, the A-weighted level for that angle is 5 to 10 dBA below the level at the 170 degree location.

Test personnel performed noise surveys during quiet periods when the background noise was minimal, e.g., early in the morning when no other aircraft or engine test stands were operating.

Volume 2 of the handbook describes the influence of meteorology on far-field noise environments, and provides, if required, the factors necessary to adjust the handbook's standard meteorological day data.



CONCRETE SLAB

Figure 2. Far-Field Measurement Locations on a Taxiway at Eglin AFB, FL

	ON							OMEGA 3.2
NOISE SOURCE/SUBJECT:	÷.	OPERATIONS	O			~ ^		RUN 01
OV-10A AIRCRAFT GROUND CREW NEAR FIELD NOISE) 26 JUL 76) PAGE F1
	1/4	2/8	3/6	7 0/4	OCATIC 5/8	LOCATION/CONDITION 5/8 6/8 7/8	DITION 7/8	8
FRED (HZ)								
31.5	83	9.0	66	93	8.9	93	87	
63	95	103	113	104	105	107	96	
125	87	101	114	104	102	103	100	
250	95	103	115	107	106	105	101	
500	93	103	119	110	101	105	66	
1000	84	96	110	104	101	16	96	
2000	19	104	101	110	103	103	76	
0004	82	110	110	105	101	106	93	
8000	79	102	113	106	103	100	88	
OVERALL	86	113	123	116	114	113	106	

NOTRE COURSE/CITA IECT+) OMEGA 3.2
יחושה שמחשים של שניין	COPE	OPERATION				~) RUN 01
OV-10A AIRCRAFT											26 JUL 76
NEAR FIELD NOISE											PAGE H1
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OASLA	101 3	113	119 p	2.2	1111	110	102				
MINIMUM QPL EAR MUFFS											
OASLA*	74	287	98	91	68	88	83				
AMERICAN OPTICAL 1700 EA	FF		2	2	1	,	;				
OASLA*		83	66	98	9.4	84	7.8				
20110 643 013-7	960	571	7.1	339	480	480	196				
OASLA*	2.0	85	96	80	87	9.4	7.8				
	7 196	404	60	240	285	480	960				
PTICAL 1700	FFS	PLUS	V-51R	EAR	PL UGS						
OASLA*		68	83	73	73	69	49				
H-133 GROUIND COMMUNICATION UNIT	960 g	090	096	960	196	960	096				
OASLA*	64	82	68	85	83	81	47				
_		629	202	404	571	807	096				
COMMUNICATION PREFERED SPEECH INTER PSIL	H INTERFERENCE LEVEL 85 102	LEVEL 102	(PSIL	N 6	08)	102	96				
ANNOYANCE PERCEIVED NOISE LEVEL	(PNL IN	IN PNDB)			č	,					

* BASED ON CALCULATED SPL SPECTRUM UNDER PROTECTIVE DEVICE. P ADDITIONAL EAR PROTECTION REQUIRED.

TABLE 4

TEST CONDITIONS

FOR FAR-FIELD NOISE MEASUREMENTS

OV-10A Aircraft, Ground Runups, Eglin AFB, FL 10 February 1969 Tail # 6613553,

Aircraft Engine Operation

Idle
Both Engines
70 % RPM
600 foot pounds torque

Locked Props 89 % RPM < 600 ft. lb torque

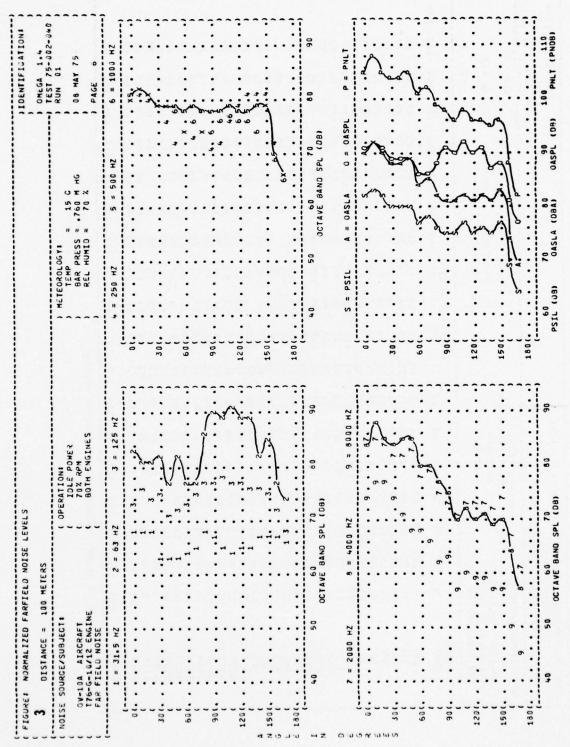
Military Power 101 % RPM 1900 ft. lb torque

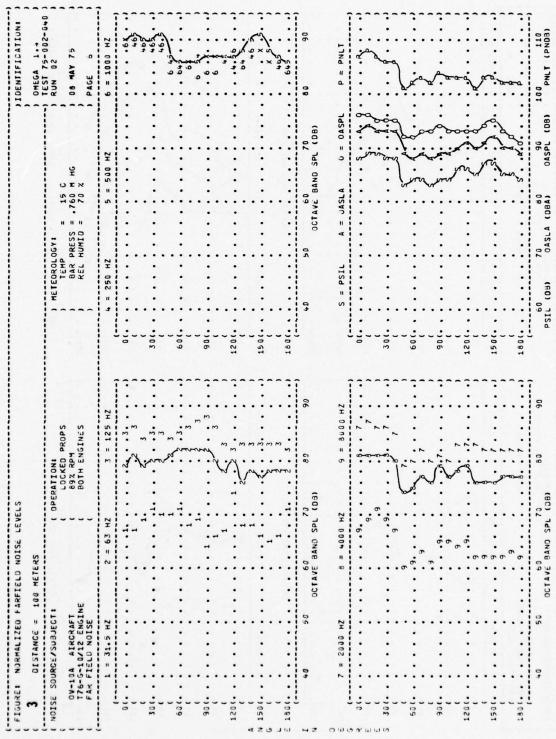
Meteorology

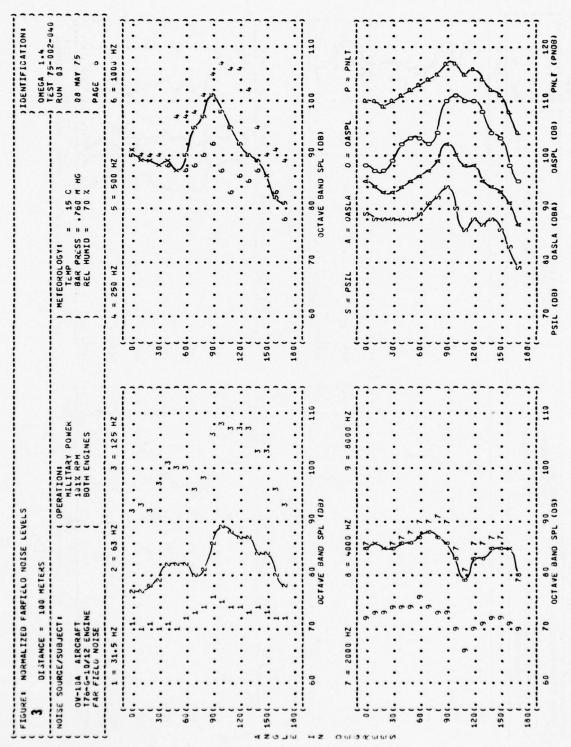
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250			12	72	22	20	6 6	29	29	1 6	90	9 8	7.7	7.7	72	72	72	99	20	
315	80			81	19	16	72	73	73	73	7.1	69	15	81	83	83	82	73	65	
00+	7	78 7		82	142	15	73	73	14	12	7.1	7.1	14	75	7.8	78	78	69	19	
200	80	0		92	22	18	83	19	8.0	7.8	78	4	80	11	14	11	92	69	9	
630	7			62	11	25	14	72	73	72	14	11	14	14	15	91	92	99	63	
900	1			62	16	11	22	72	73	72	72	75	11	14	14	7.1	92	99	63	
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1600	8	81 8		62	18	11	78	72	15	72	7.1	69	72	72	73	7.1	14	29	61	
2000	80			82	62	42	80	22	92	73	7.1	69	72	7.1	7.1	20	72	65	29	
2500	8	ıs		94	82	82	83	7.8	62	15	7.1	69	7.1	69	69	69	69	49	25	
3150	•			94	83	9.4	* 8	29	62	16	72	69	7.1	69	69	68	69	63	26	
0004	8			94	82	94	85	80	80	11	15	2.0	72	7.0	7.0	68	69	63	96	
2000	2	8 22		82	92	80	80	92	92	7.1	7.1	99	99	65	99	49	49	28	25	
6300	2			11	92	72	20	6.8	29	62	49	66	62	60	61	28	28	25	94	
8000	1			62	73	72	6.8	20	6.8	49	9	25	66	28	28	99	55	64	t	
10000	9			20	69	69	62	62	09	9	29	52	25	21	25	24	53	24	41	
OVERALL	6	13 9	2	93	95	91	92	89	68	91	93	93	46	93	93	68	91	63	79	

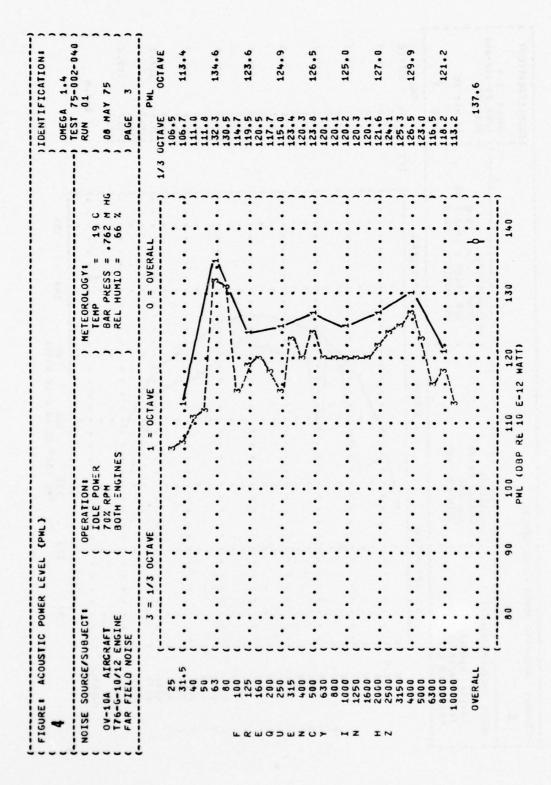
TABLE: 5	MEASURED 1/3 OCTAV DISTANCE	SOUN E BA	D PRES ND 76 MLT	RESSURE	PRESSURE LEVEL	(03)											DENTIFICATIONS OMEGA 1.4	1CATI	
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176-6-1	176-6-10/12 ENGI	INF			BOTH	PNGINE	0				RFILE	HUMID	:		2	-		•	
	FAR FIELD NOISE												,	,		-	PAGE	2	
FREG								Z	GLE (DEGREES	ES)								
(HZ)	0	10	20	30	0,	20	09	7.0	8.0	06	100	110	120	130	140	150	160	170	180
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31.5	99 9		69	7.0	7.1	69	7.1	29	69	49	99	29	99	69	6.8	29	63	9	69
0+	99		67	7.0	29	63	68	89	99	61	61	63	92	69	49	62	63	9	69
20	69		62	69	63	63	49	63	63	61	63	99	73	20	49	63	63	49	99
63	20		29	29	99	2.0	99	68	69	99	68	7.1	9.0	20	29	29	29	67	69
980	91	80	85	93	80	83	4 0	0 0	82	10	82	80	28	11	80	80	00	80	62
100	01	22	6 5	2 4 5	* *	9 6	90	20	100	9 6	96	32	62	9	29	63	29	20	9 6
160	87	87	85	85	85	8 2	85	t t	200	96	9 4	83	82	20 00	83	81	82	9.6	82
200	9.6	83	83	83	83	8.0	82	65	96	88	19	87	94	40	83	81	80	81	90
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630	89		89	60	6 6	98	85	82	9 7	82	87	87	87	96	84	89	82	82	82
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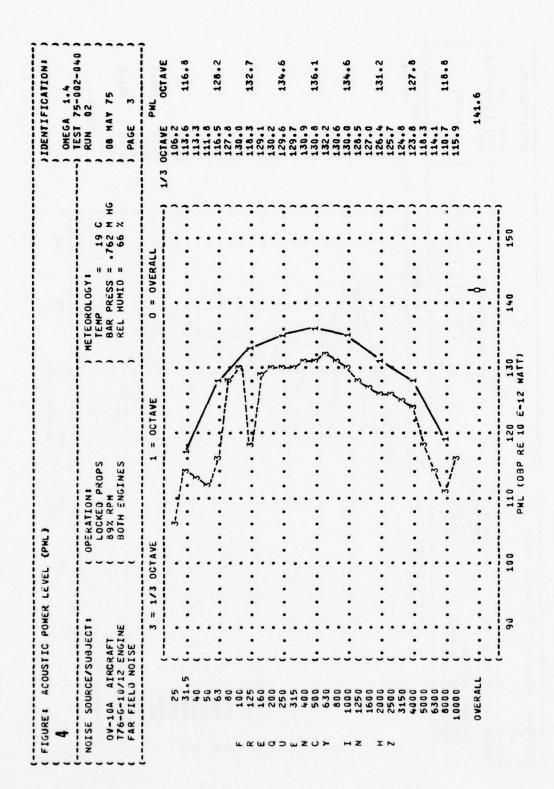
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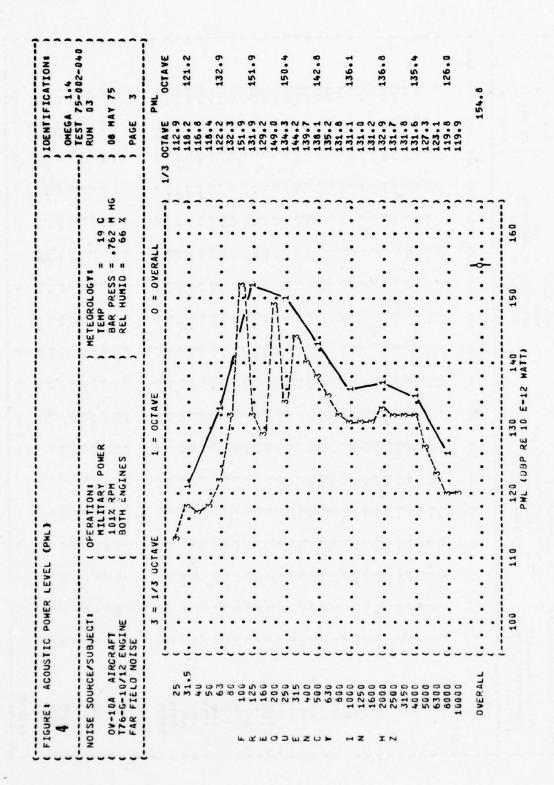








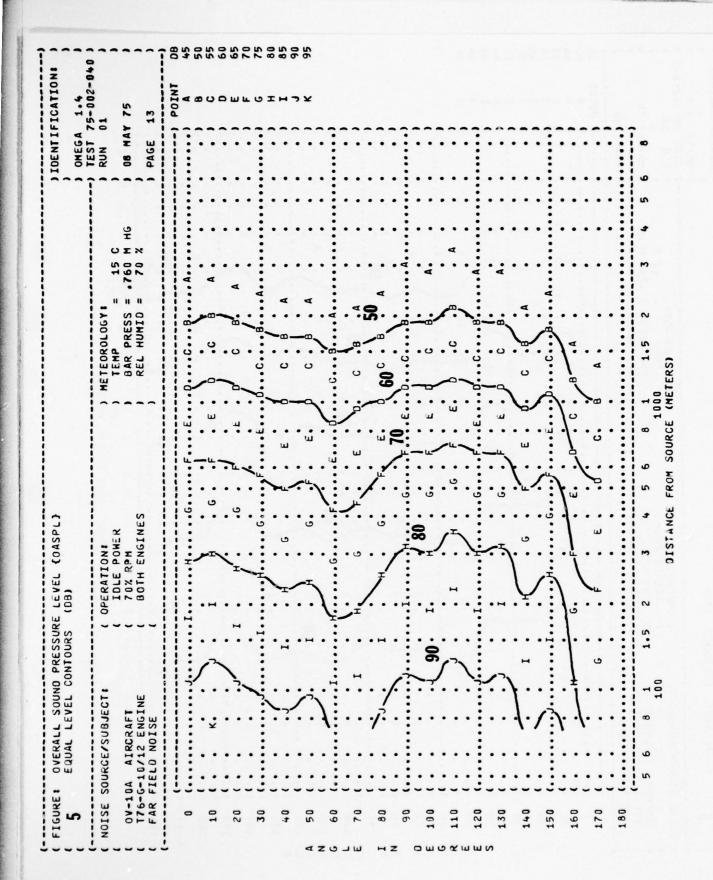


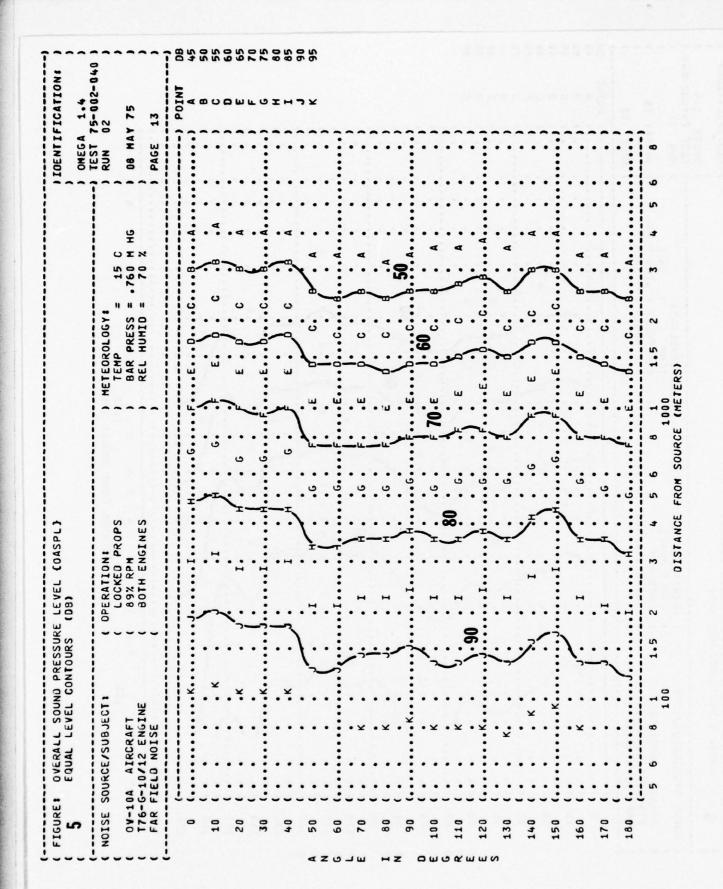


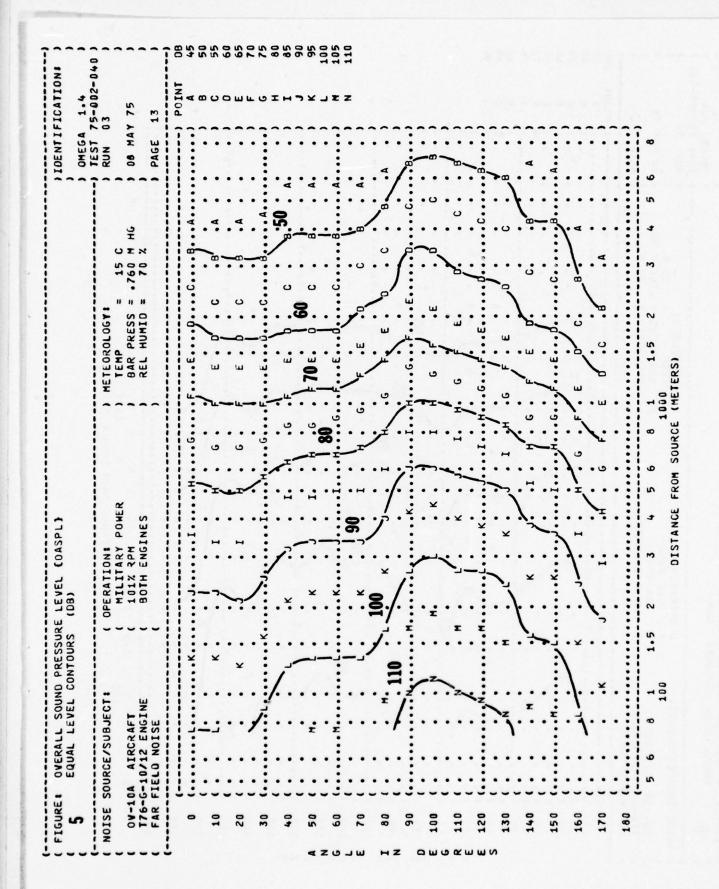
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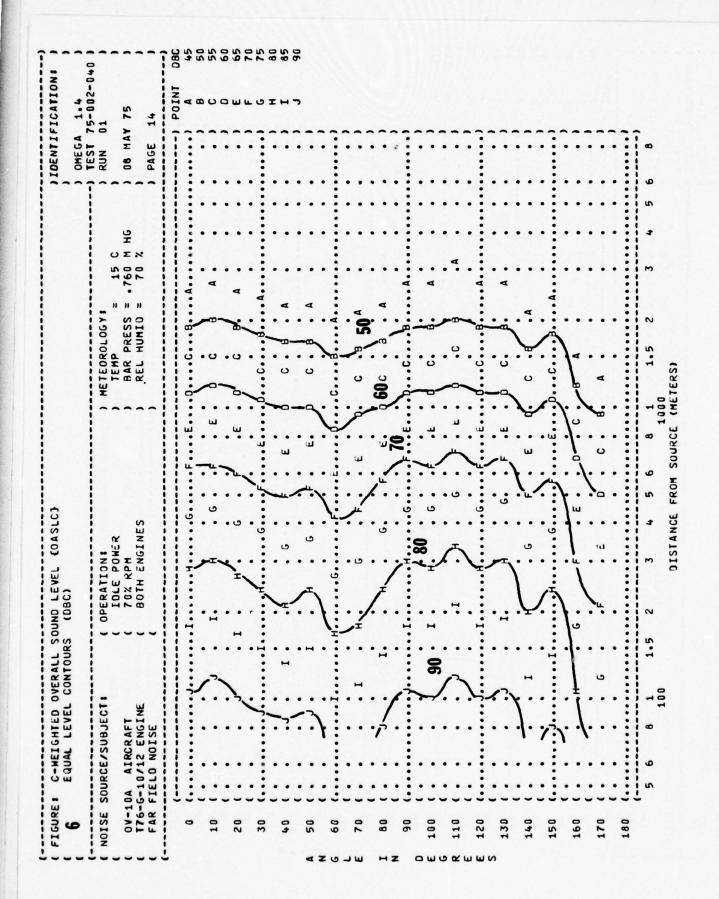
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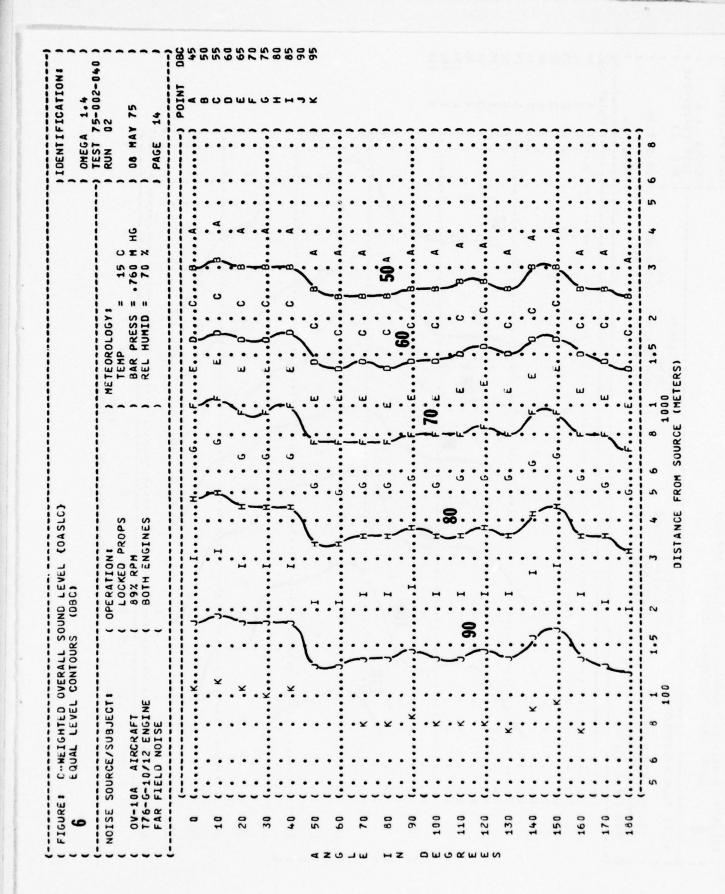
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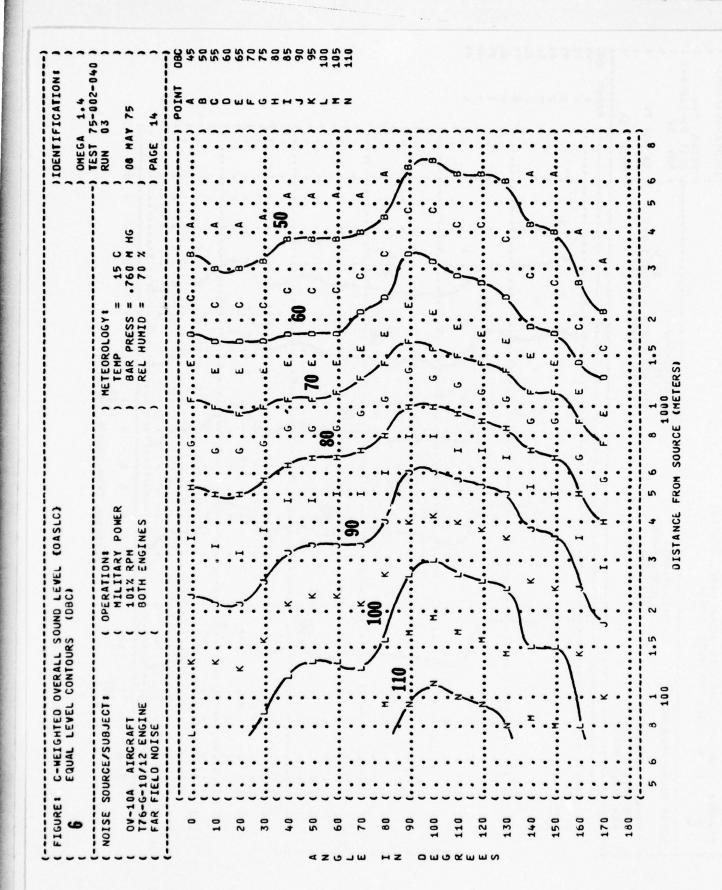


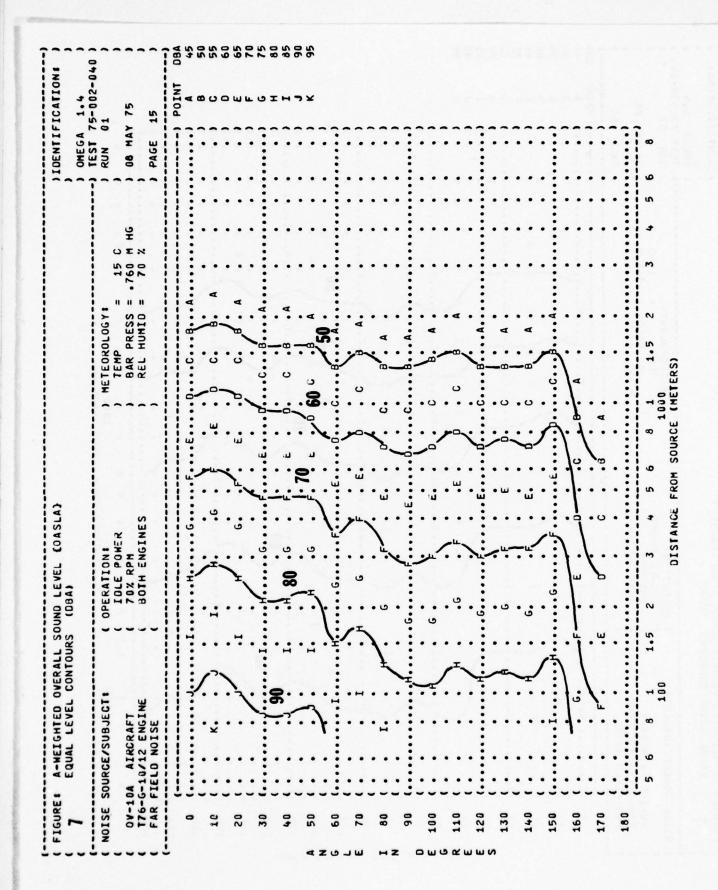


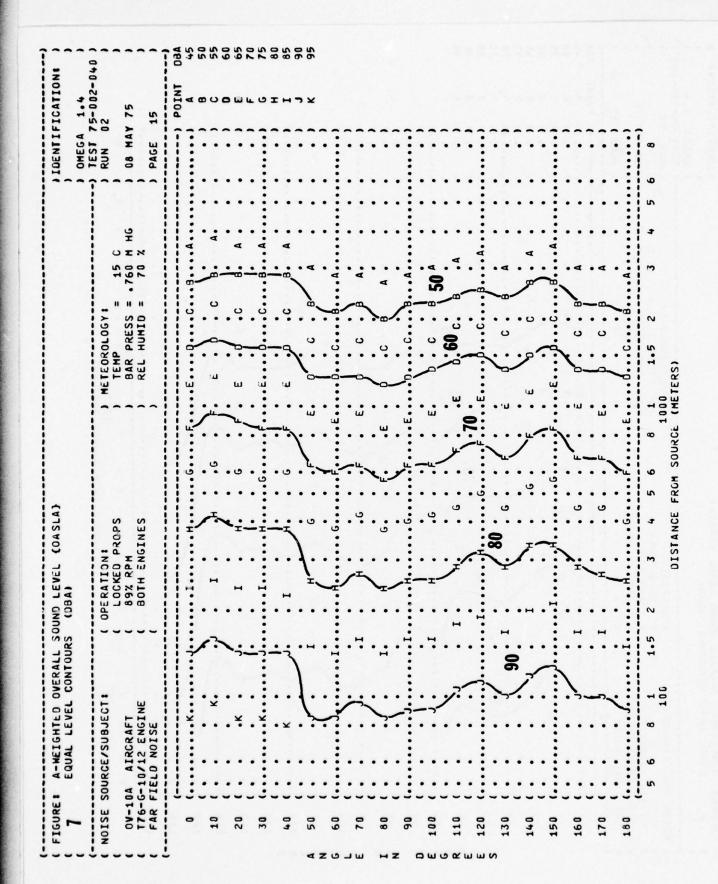


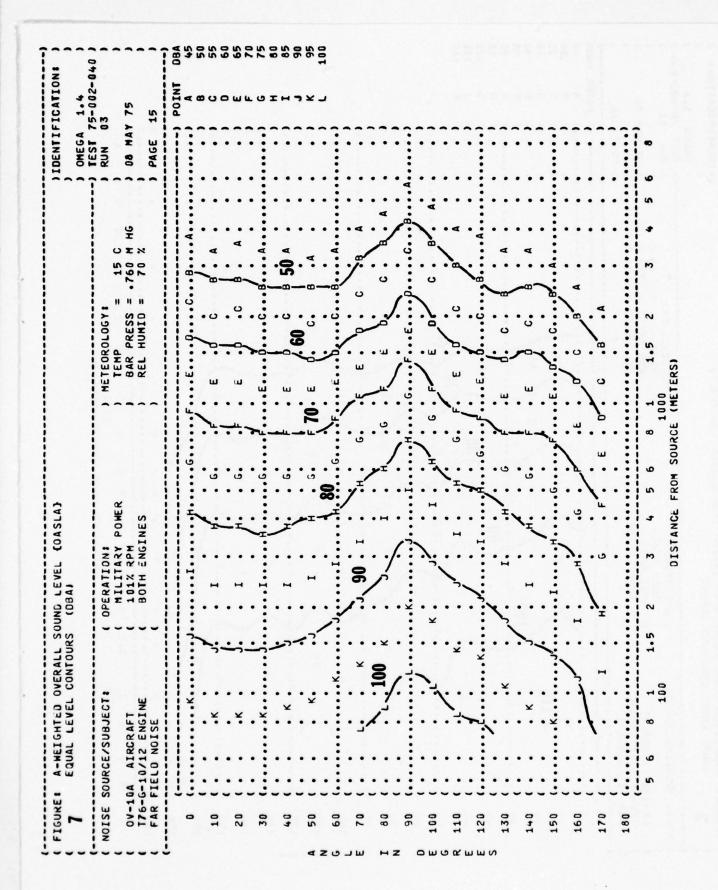


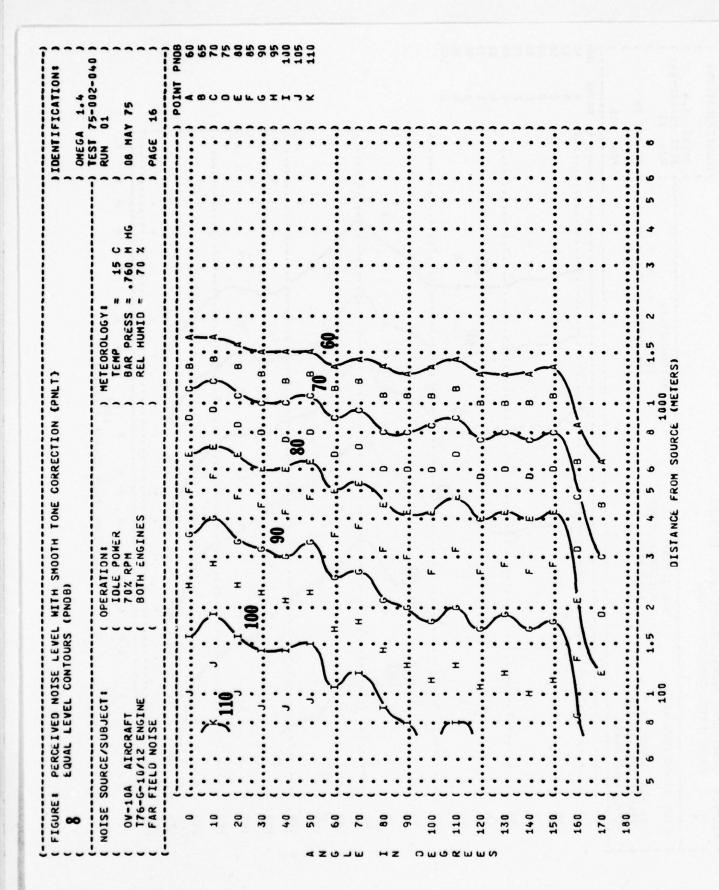


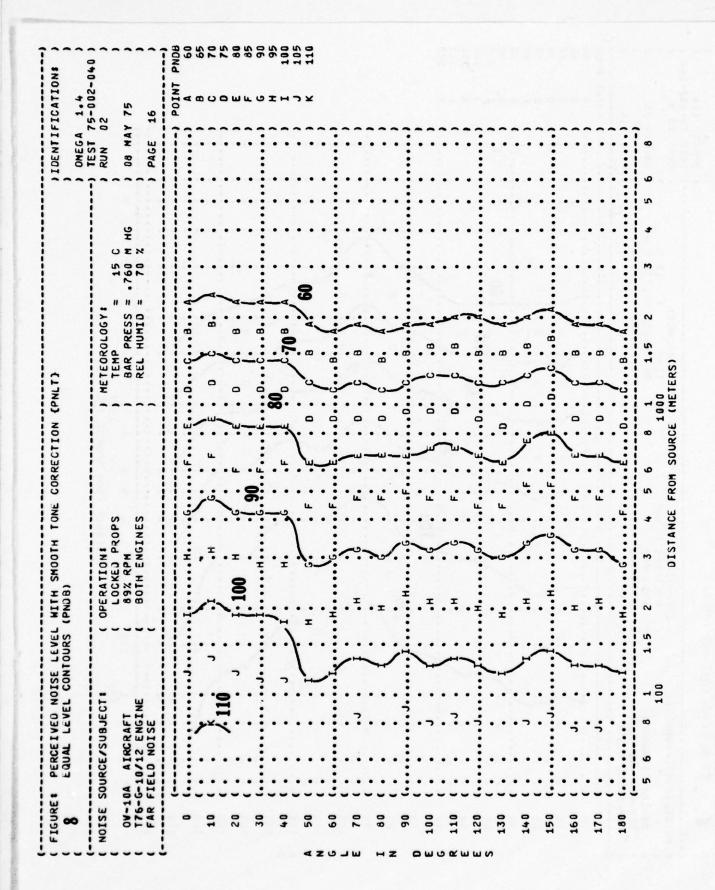


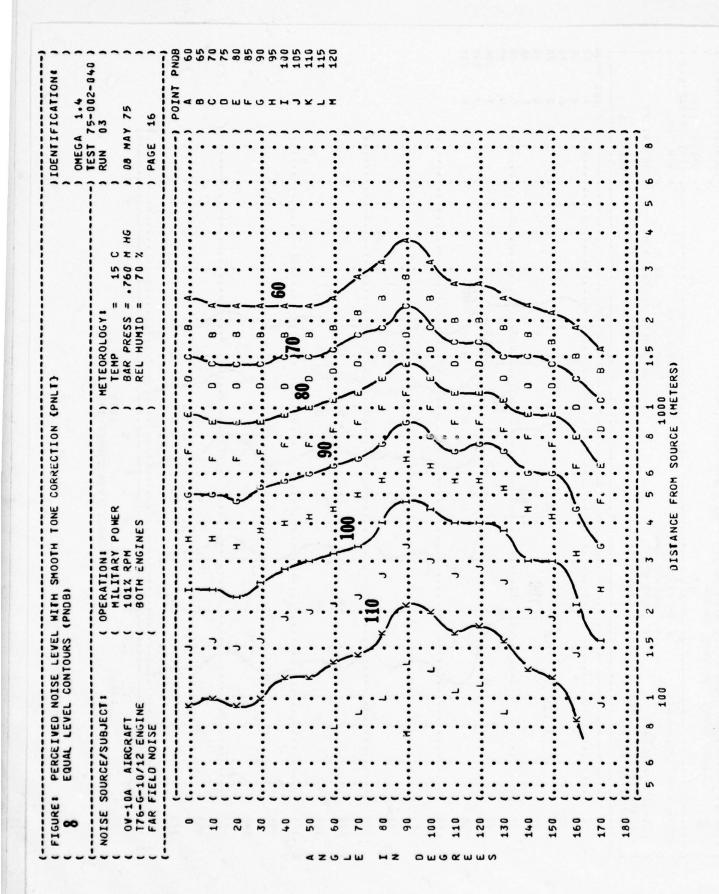


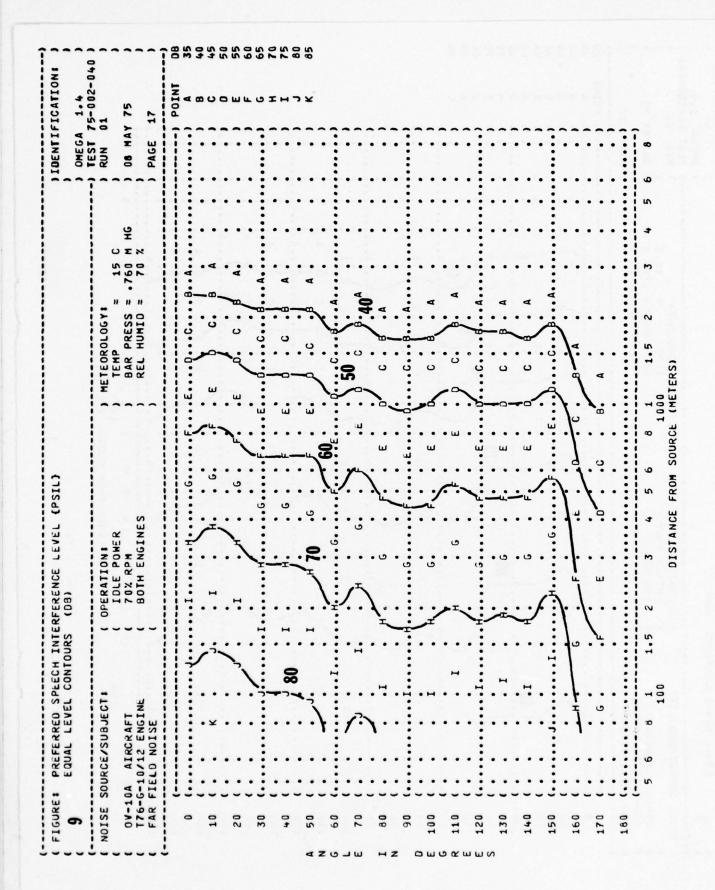


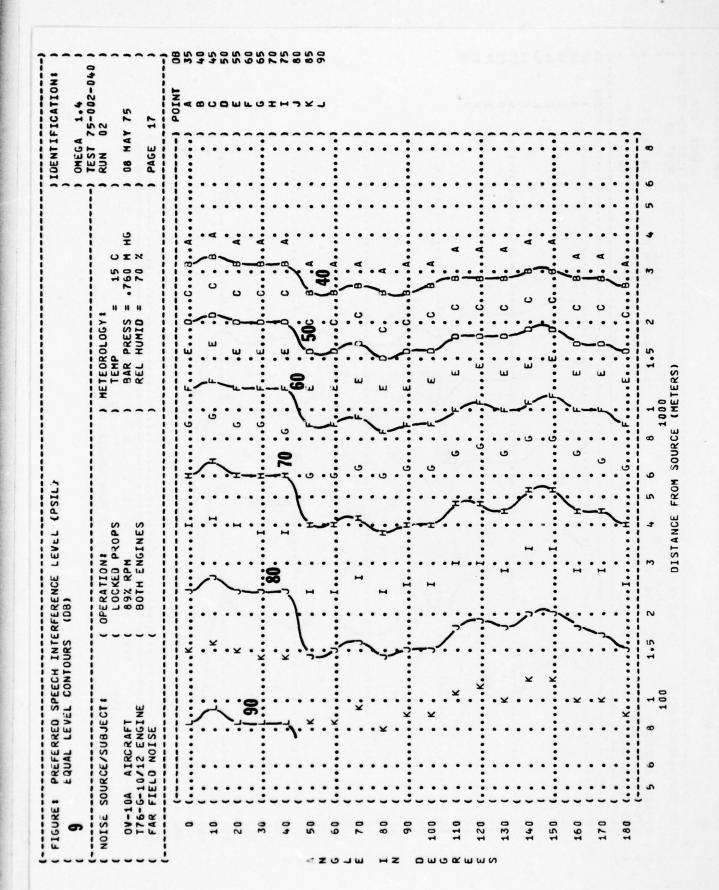


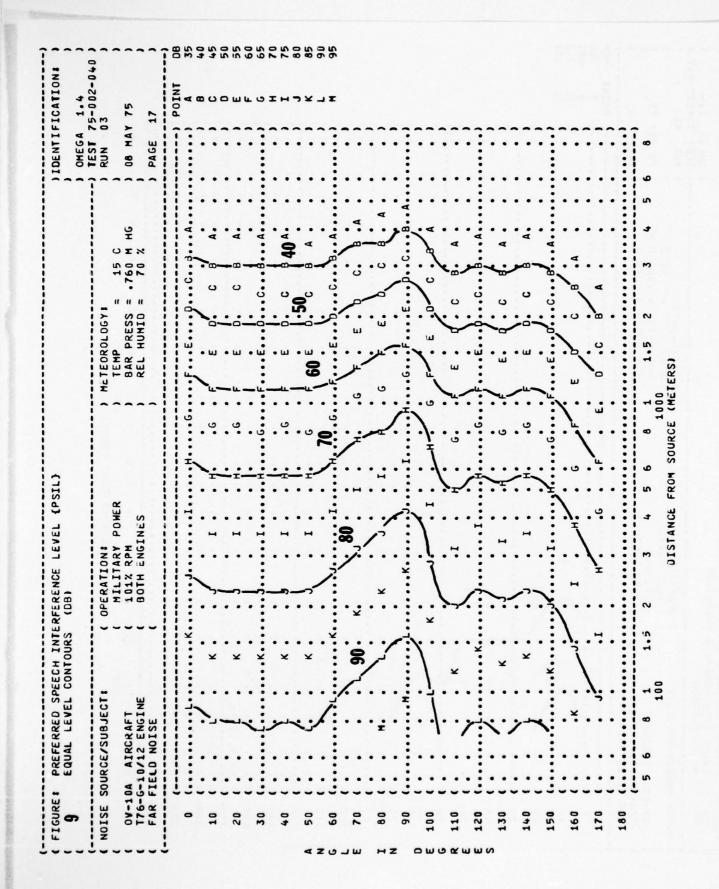












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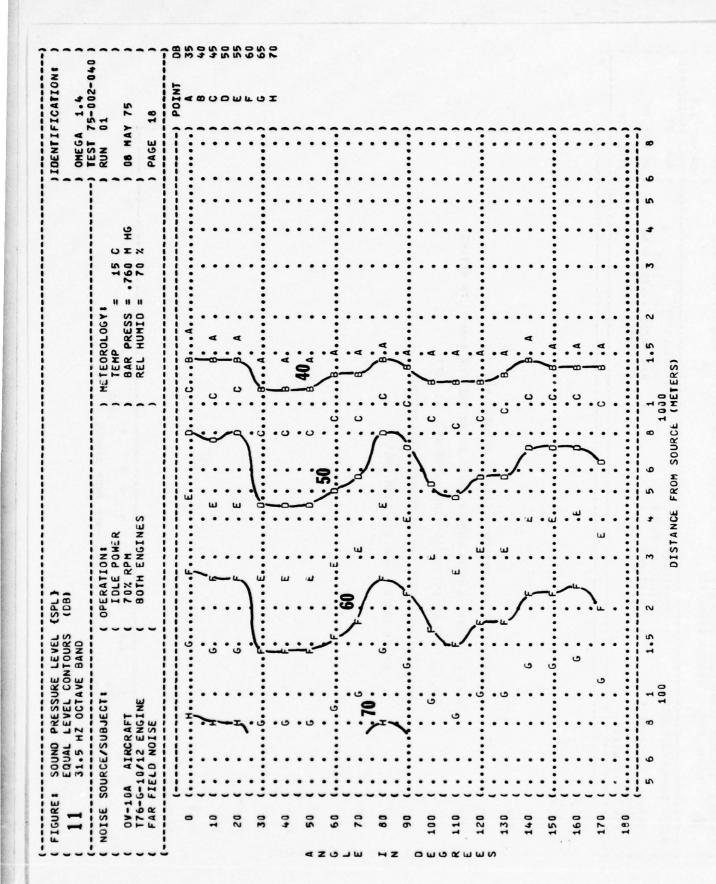
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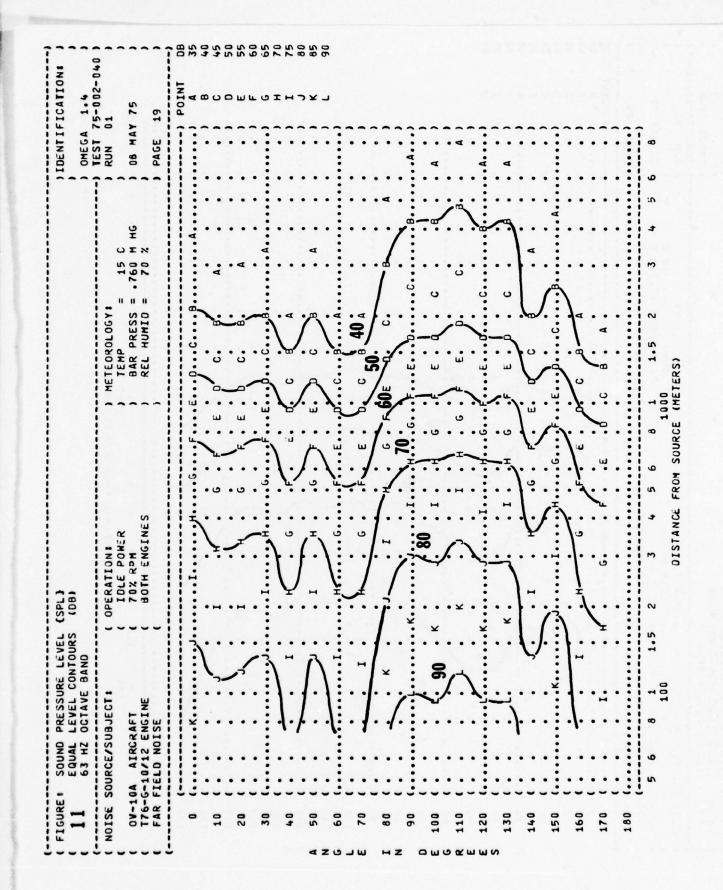
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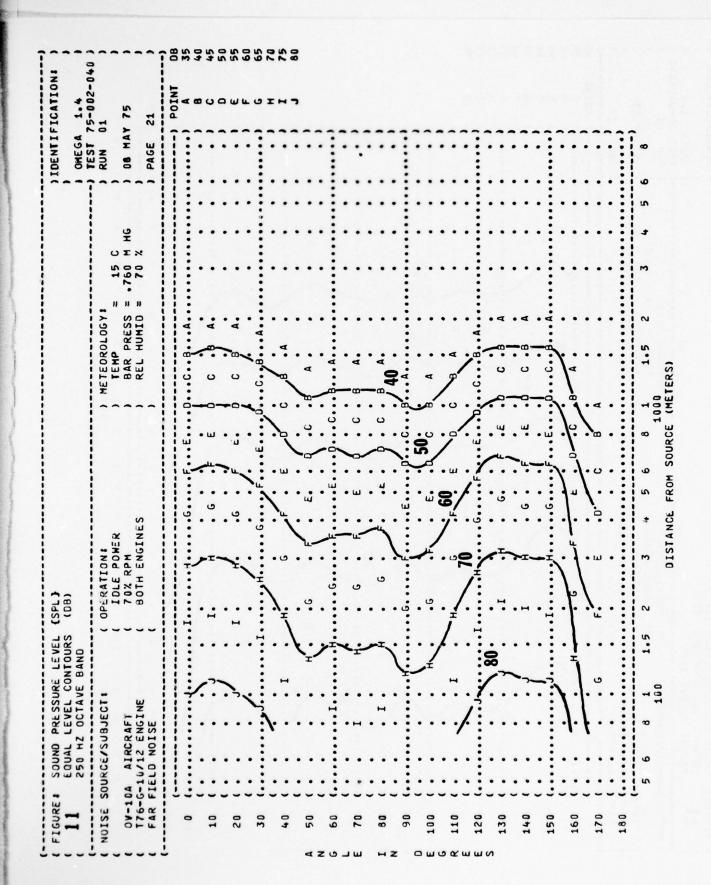
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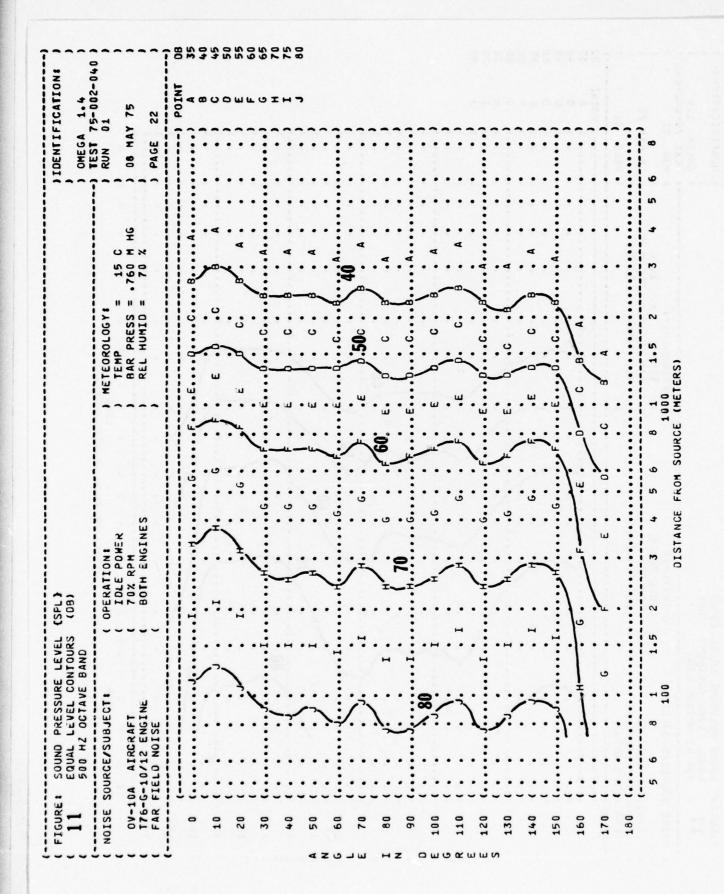
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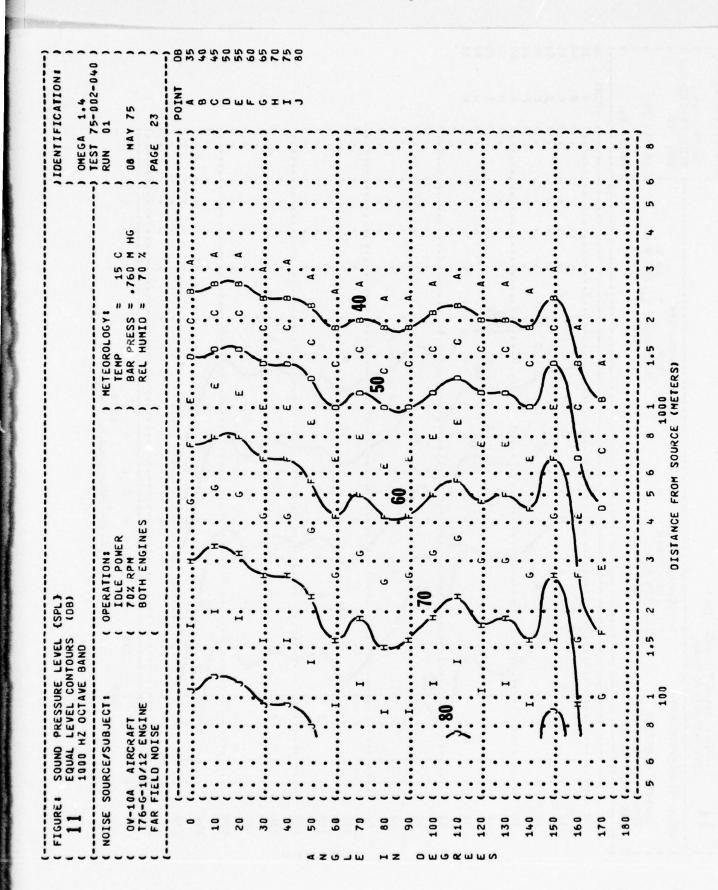


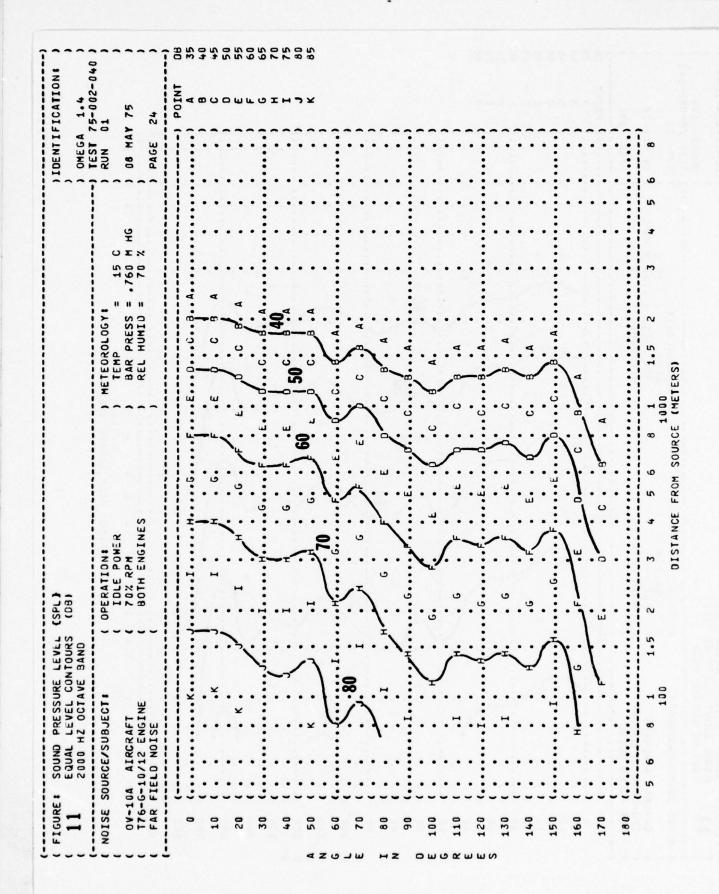


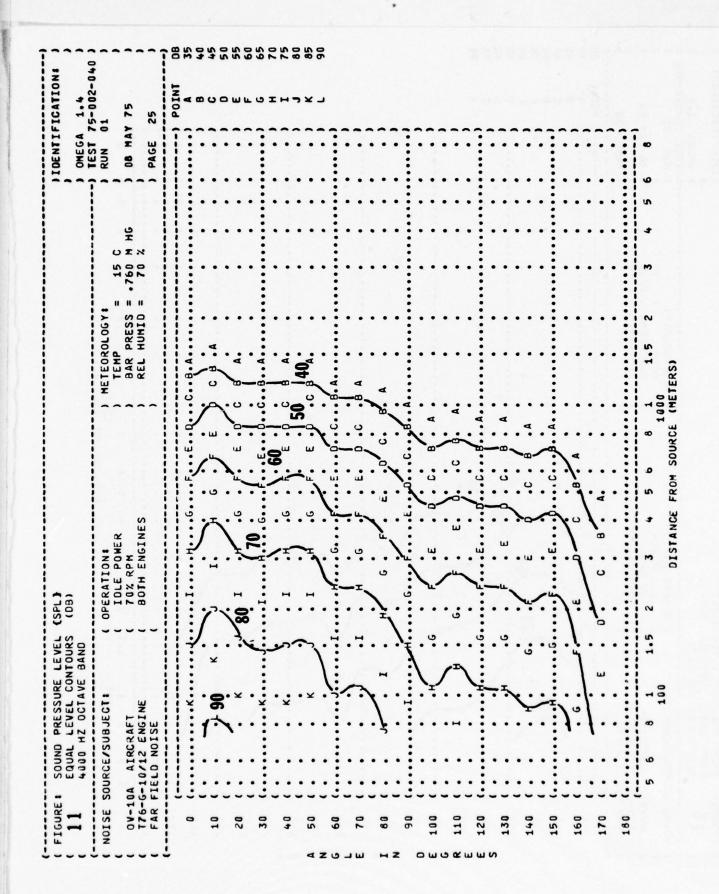
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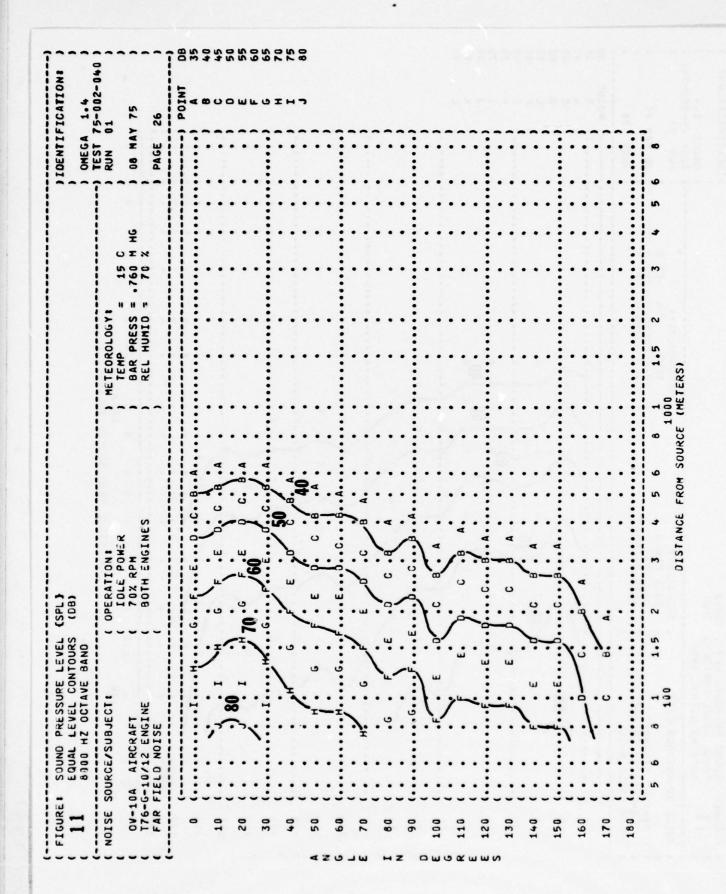


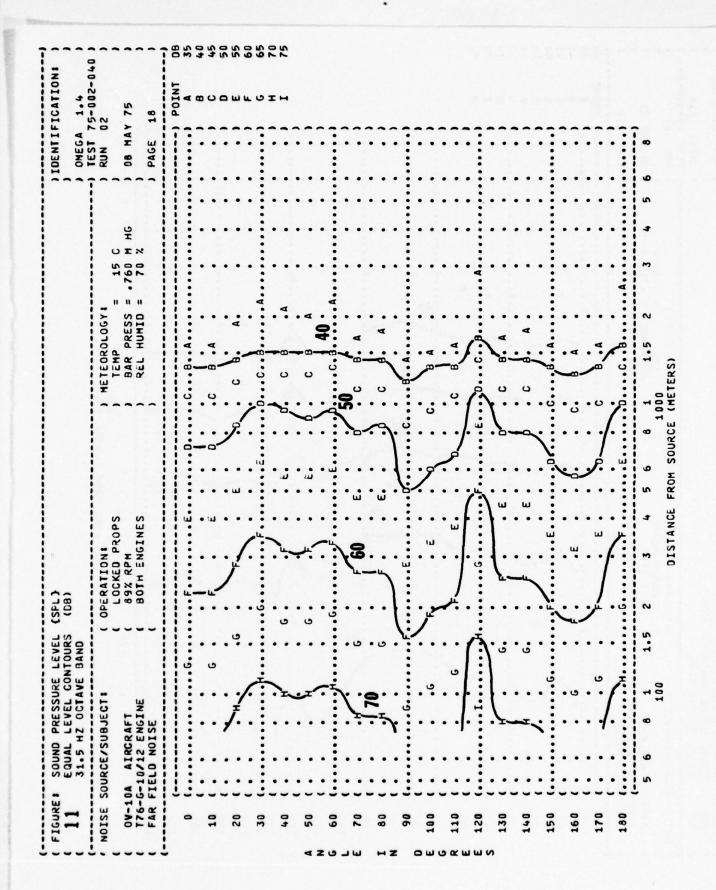


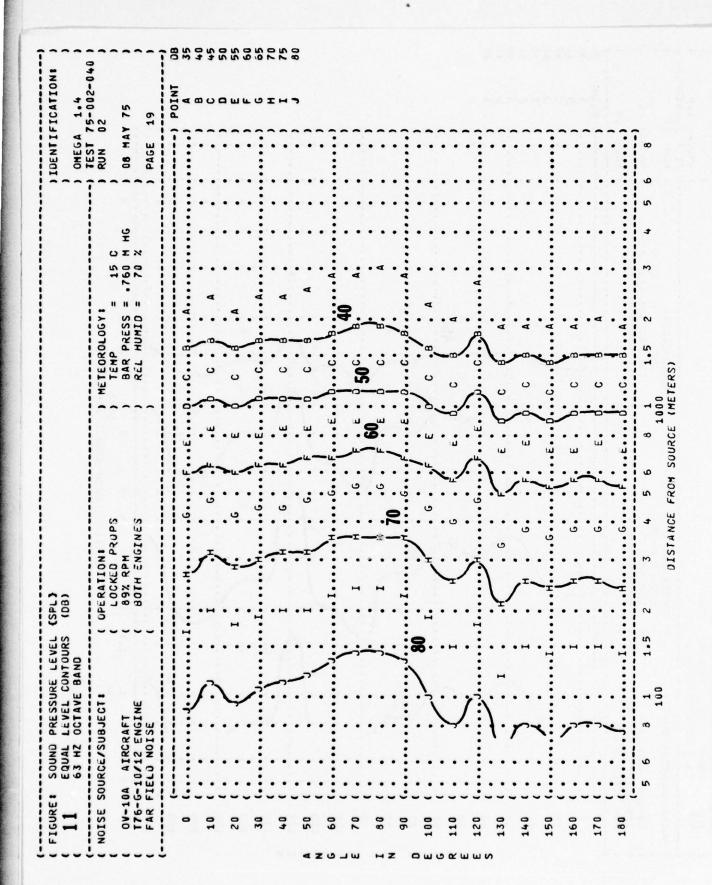


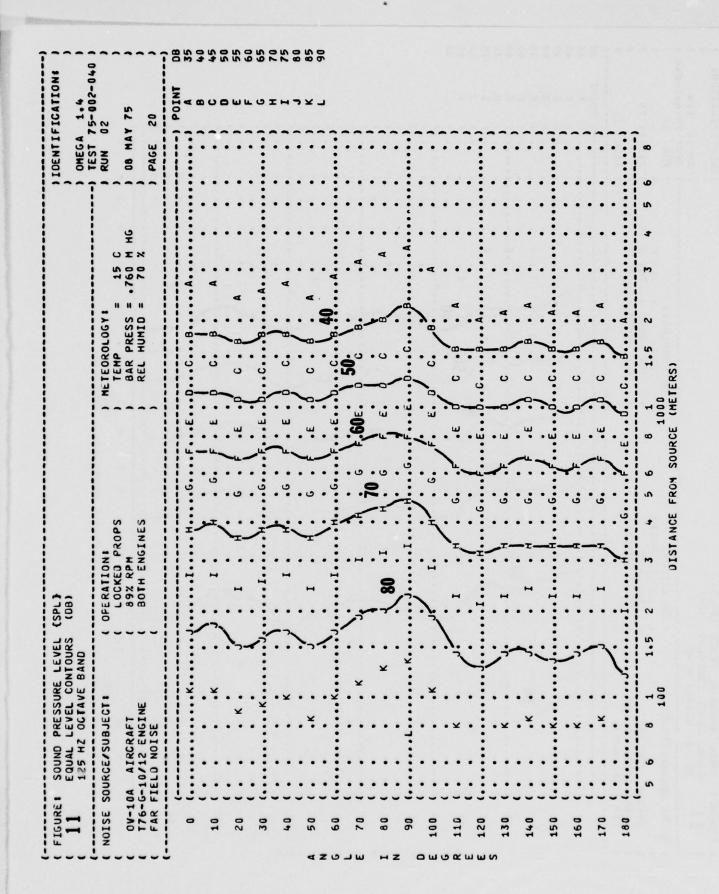


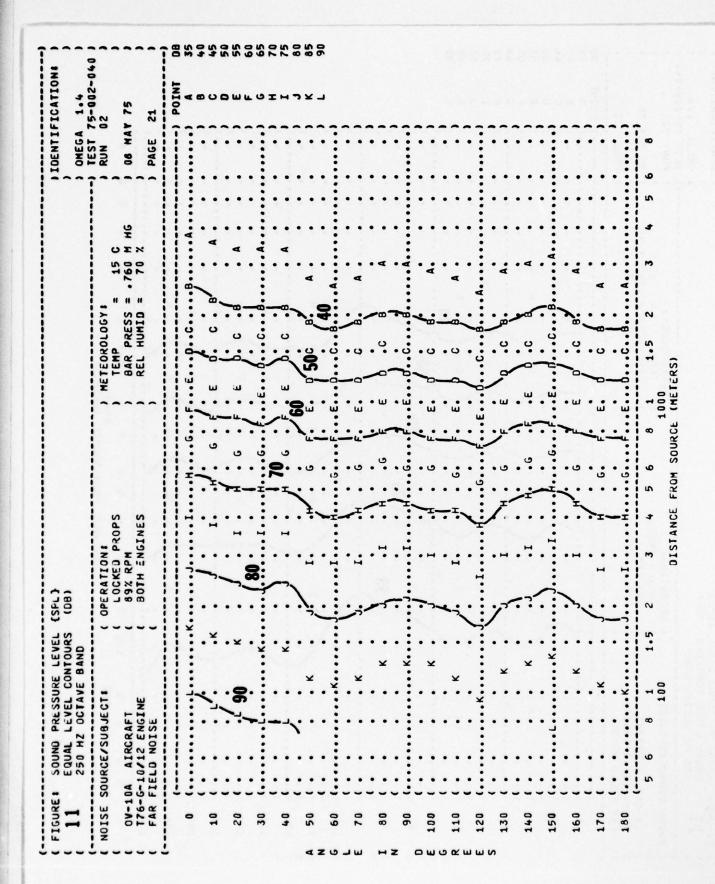


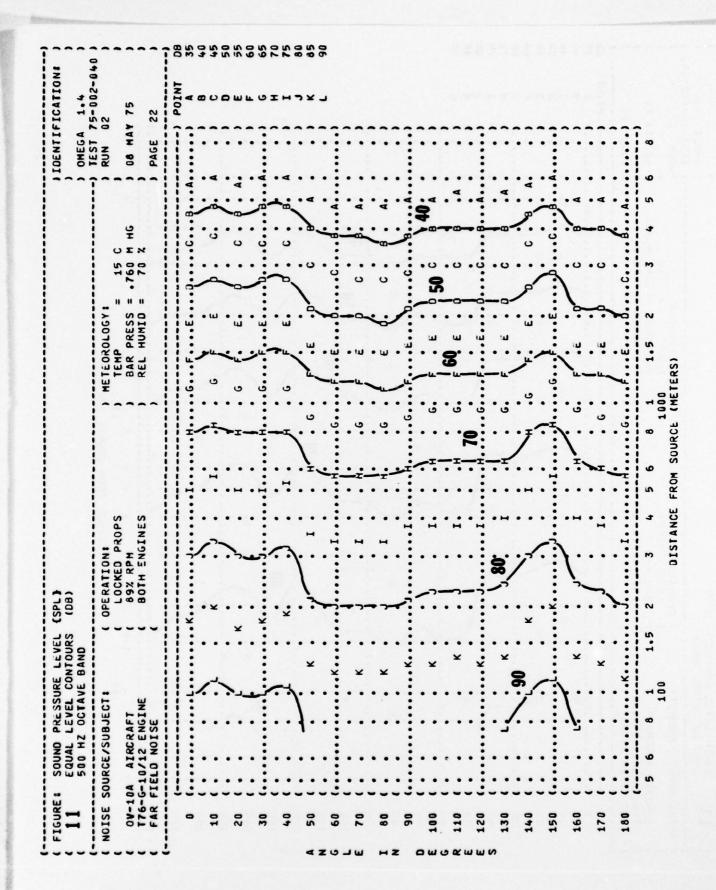


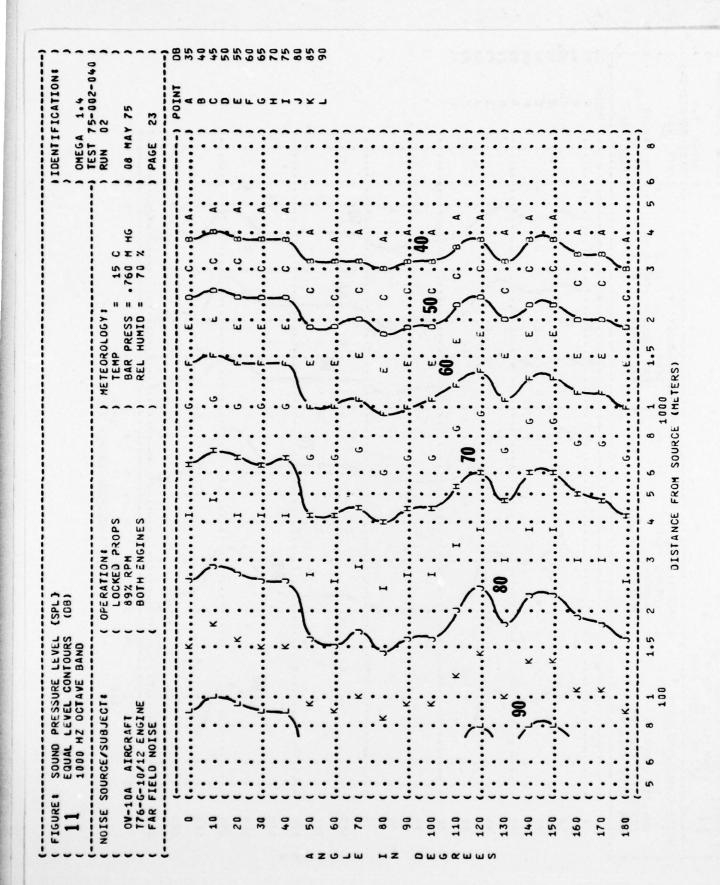


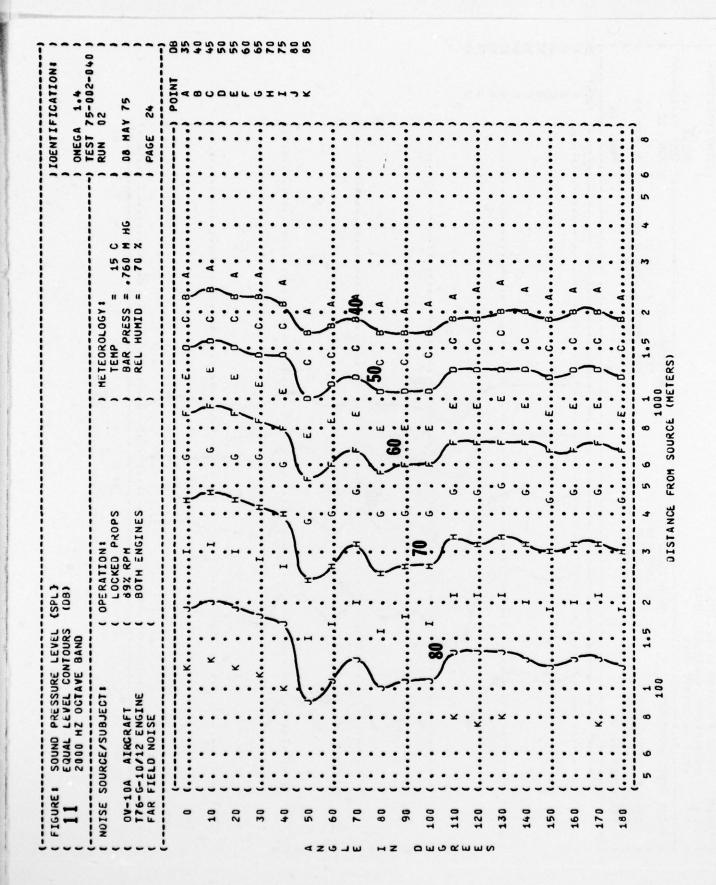


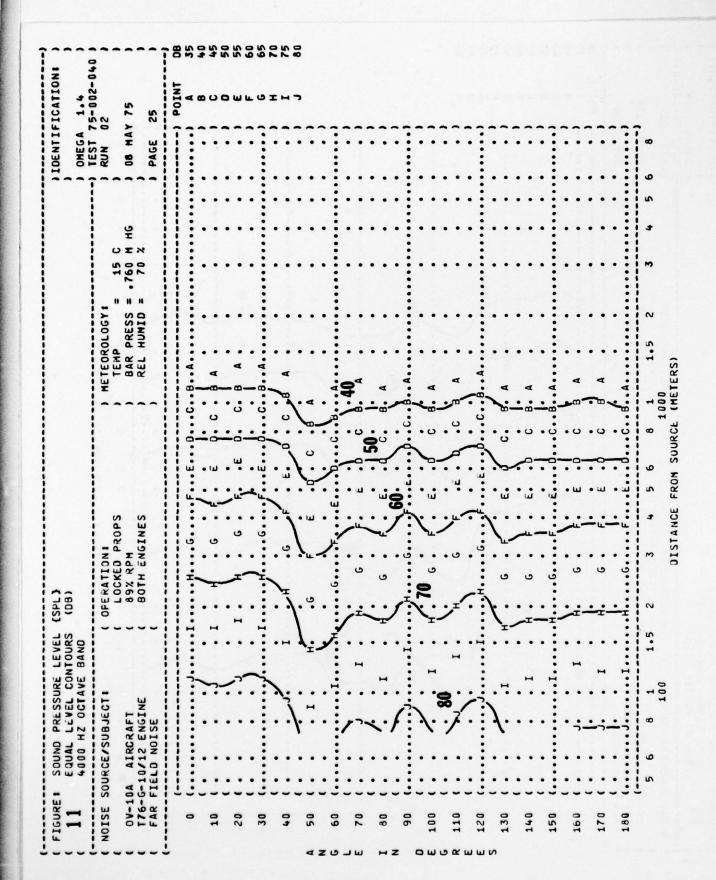












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